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Investigation on Coupling Effects between Surface Wear and Dynamics in a Spur Gear System

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Abstract: A dynamic wear prediction methodology is proposed to investigate the coupling effects between surface wear and dynamics of spur gear systems. The overall computational scheme combines a quasi-static wear model and a translational-rotational-coupled nonlinear dynamic model. The worn surfaces are represented by modulated mesh excitations and introduced into the dynamic model to investigate the effects of surface wear on system's dynamic characteristics. The dynamic gear mesh forces are converted into an equivalent load by using the Miner rule to reveal the effects of dynamics on wear behaviors. A spur gear transmission is taken as an example system to demonstrate the interactions between surface wear and dynamic behaviors. The simulations indicate that surface wear and gear dynamics are highly interacted.

Keywords: gear dynamics; gear wear; mesh stiffness; transmission error

1. Introduction

Gear systems are widely used in various power transmission applications due to their distinguished merits of accurate transmission ratio, large power range, high transmission efficiency and stable operation quality. For a gear system worked in dirty environments with heavy duties, the surface wear and the vibration are considered two major causes of operational failure. To improve the transmission accuracy and avoid premature fatigue, numerous efforts have been carried out to investigate the vibration mechanism of gear systems [1-7] and the wear mechanism of gear teeth [8-32].

As addressed in previous studies, surface wear and gear vibration are mutually affected by each other [8-17]. As a long-term material removal behaviour, surface wear results in a deviation from the intended tooth profile and alters the gear mesh excitations, which makes the dynamic behaviors of a gear system with worn surfaces quite different from their counterparts without wear. Surface wear not only alters the stress and load distributions but also aggravates system's characteristics of vibration and noise. On the other hand, the dynamic loads caused by system vibration are different from

A_e, B_e	Magnitudes of AM and FM for STE	m_p, m_g	Mass of pinion and gear
A_k, B_k	Magnitudes of AM and FM for mesh stiffness	n	Number of harmonic terms
F_{aT}	Fluctuating force	n_1	Speed of pinion
F_{aTr}	Harmonic amplitude of fluctuating force	p	Contact pressure
F_m	Average mesh force	q	Number of pressure update
F_p, F_g	External radial preloads of bearings	s	Sliding distance
I_p, I_g	Mass moments of inertia for pinion and gear	t	Time
R_p, R_g	Radii of base circles for pinion and gear	u	Relative displacement along the line of action
T_{eq}	Equivalent load	y_p, y_g	Transverse displacements of pinion and gear
T_j	Torque at each discrete time instant	δ	Displacement in the direction of the path of action
T_p, T_g	Input torque and output torque	$\delta_{H1,2}$	Hertz flattening
$2b$	Backlash value of gear pair	δ_{RK}	Gear body deformation
c_m	Mesh damping	δ_Z	Gear tooth bending
c_{py}, c_{gy}	Damping of bearings for pinion and gear	ε^q	Predetermined wear threshold
e	STE	ε^t	Maximum allowable wear threshold
e_r	R -th harmonic amplitudes for STE	ζ	Number of wear cycles
h	Wear depth	θ_p, θ_g	Torsional displacements of pinion and gear
k	Wear coefficient	φ_{aTr}	Harmonic phase angle input torque excitation
k_{ar}	R -th harmonic amplitudes for mesh stiffness	$\varphi_{ea}, \varphi_{eb}$	Initial phases of AM and FM for STE
k_m	Time-varying stiffness	φ_{er}	Harmonic phase angles of STE
k_{mm}	Average mesh stiffness	$\varphi_{ka}, \varphi_{kb}$	Initial phases of AM and FM for mesh stiffness
$k_{t1,t2}$	Total stiffness of one tooth pair	φ_{mr}	Harmonic phase angle of mesh stiffness
$k_{p,g}$	Individual stiffness of one tooth	ω_{aT}	Fundamental frequency of input excitation
k_{py}, k_{gy}	Bearing stiffness of pinion and gear	ω_e	Fundamental frequency of STE
l	Slope of the Wöhler-damage line	ω_m	Fundamental frequency of mesh stiffness
m_c	Equivalent mass of gear pair		

the static loads, which increases the contact pressures between the mating surfaces and fasten the wear process. From this point of view, the gear system dynamics also affects the surface wear characteristics significantly.

Although the investigations on surface wear and gear dynamics are abundant, the studies focused on the their coupling effects are quite few. Among these few studies, most of the them are concentrated on the effects of surface wear on the dynamics of a gear system [8-14]. For example, Choy et al. [8] studied how the surface pitting and wear affect the vibrations of a gear system. The effect of surface wear was represented by the change of gear mesh stiffness which was then introduced into a gear-shaft model. Wojnarowski and Onishchenko [9] established a two degrees-of-freedom elastic dynamic model with worn teeth to study the impacts of surface wear on spur gear dynamics. Yesilyurt et al. [10] investigated the influence of surface wear on the change of gear mesh stiffness from the durability and diagnostic points of view. Their study manifested that the effect of surface wear on the system's dynamics was primarily caused by the

deviations in tooth shape from the idea involute profile. Kuang and Lin [11] investigated the effect of surface wear on the variation of vibration spectrum of a spur gear system by combining a single degree-of-freedom dynamic model with a wear prediction model developed by Flodin et al. [18,19]. The simulation results showed that the surface wear may change the dynamic load histogram of an engaging spur gear pair greatly. Later, they carried out a similar investigation for an engaged plastic gear pair [12]. However, in their studies only the dynamics at one rotational speed were investigated. Yuksel and Kahraman [13] combined a wear prediction model proposed by Bajpai et al. [23] and a deformable-body dynamic model [33] to study the influence of quasi-static wear profiles on the dynamics of a planetary gear set. The worn surfaces were introduced into the dynamic model to quantify the effect of surface wear on the dynamic gear meshing forces. Nevertheless, the surface wear behaviors under dynamic conditions were not discussed in this study. Osman and Velex [14] combined a dynamic model with a wear model based on Archard's law to investigate the effect of surface wear on the dynamic responses of a wide-faced gear.

Compared with the studies of the effects of surface wear on gear dynamics, the investigations on the effect of gear dynamics on the surface wear behaviors were even less [16,17]. Ding and Kahraman [16] combined a one degree-of-freedom dynamic model with Bajpai's wear prediction model [23] to study the interactions between the surface wear and the dynamics in a spur gear system. In this study, the effects of wear profiles were represented by a periodically time-varying mesh stiffness function and an external displacement excitation. More recently, they extended the methodology to a planetary gear system to investigate the interactions between the surface wear and the system's dynamic behaviors [17].

From the above reviews, it can be found that there is common thread in the efforts of investigating the coupling effects between the surface wear and gear dynamics in that the worn surfaces are treated as internal excitations (time-varying meshing stiffness and/or transmission error) incorporated with certain kind of dynamic model of a gear system. Based on these dynamic models, the effects of surface wear on the dynamic behaviors of gear are investigated and vice versa. However, it needs to point out that an accurate prediction for the gear dynamics relies on a comprehensive dynamic model. Unfortunately, the dynamic models used in the above studies only included the torsional

deflections in gear-shaft systems. The translational effects coming from the shaft bending and bearing radial deflections were not considered in these dynamic models, which may degenerate the prediction accuracy of the dynamic analysis. To guarantee the prediction accuracy, a comprehensive translational-rotational-coupled nonlinear dynamic model with three degree-of-freedom for a spur gear system is proposed and then combined with a quasi-static wear model proposed earlier by the authors [32]. Based on this combined dynamic surface wear prediction methodology, the effects of surface wear on the dynamic behaviors of the spur gear system are investigated. Meanwhile, the influences of the dynamic loads on the surface wear behaviors are studied by converting the steady-state dynamic gear forces into an equivalent load with the Miner rule. The converted equivalent load is introduced into the quasi-static wear model to reveal the surface wear process under dynamic load conditions.

A thorough understanding of the coupling effects between the gear wear and the gear dynamics helps to prolong the service life and improve the performance of gear sets. Motivated by this thought, the main contents of this paper are arranged as follows. In section 2, a translational-rotational-coupled nonlinear dynamic model for a spur gear system is developed which incorporates the effect of worn surfaces of the mating gear pair. The profile deviations from surface wear are represented in the forms of gear mesh excitation and transmission error functions with amplitude and frequency modulations. In section 3, a dynamic wear prediction methodology is proposed by combining the proposed dynamic model with the aforementioned quasi-static wear model [32]. In section 4, a spur gear system is taken as an example to demonstrate the coupling effects between the surface wear and the dynamic responses. Finally, some conclusions are drawn in section 5.

2. Dynamic modeling for a spur gear system

In this section, an analytical nonlinear dynamic model for a spur gear system with surface wear is established, with which the dynamic behaviors of the system can be predicted.

By simplifying the real-world gear system into a mass-spring-damping system, a lumped parameter nonlinear dynamic model for a spur gear system developed by Kahraman and Singh [2] can be illustrated as shown in Fig. 1.

Herein, I_p and I_g denote the inertias of the pinion and the gear, respectively. Similarly, the mass of the pinion and that of the gear are represented with m_p and m_g , respectively. R_p and R_g illustrate the base circles of the pinion and the gear. Assume that θ_p, θ_g and y_p, y_g are the torsional and the transversal displacements of the pinion and the gear while F_p and F_g are the external radial preloads applied to the rolling element bearings. The gear mesh is described by a nonlinear displacement function $f(u)$ with time-varying stiffness $k_m(t)$ and viscous damping c_m . Meanwhile, a static transmission error (STE) of the gear pair is introduced and represented as $e(t)$. The radial stiffness and damping effects of rolling element bearings are marked as k_{py}, k_{gy} and c_{py}, c_{gy} , respectively. $T_p(t) = T_{pm} + T_{pa}(t)$ and $T_g(t) = T_{gm} + T_{ga}(t)$ are the external torques acting on the pinion and the gear, with T_{pm}/T_{gm} and $T_{pa}(t)/T_{ga}(t)$ being the constant part and the fluctuation part respectively.

As demonstrated in Fig. 1, the proposed lumped parameter dynamic model considers both the torsional and the transversal deflections in the spur gear system and the damping effects as well. By using the 2nd Newtonian law, the following equations of motion can be formulated.

$$\left. \begin{aligned} m_p \ddot{y}_p + c_{py} \dot{y}_p + k_{py} y_p + k_m(t) f(u) + c_m \dot{u} &= -F_p \\ m_g \ddot{y}_g + c_{gy} \dot{y}_g + k_{gy} y_g - k_m(t) f(u) - c_m \dot{u} &= F_g \\ m_c (\ddot{u} - \ddot{y}_p + \ddot{y}_g) + k_m(t) f(u) + c_m \dot{u} &= F_m + F_{aT}(t) - m_c \ddot{e}(t) \end{aligned} \right\} \quad (1)$$

Herein, an over dot denotes differentiation with respect to time t . u is the relative displacement along the line of action, m_c is the equivalent mass of the gear pair; F_m is the average mesh force, $F_{aT}(t)$ is the fluctuating force related to the external input torque excitation. These variables can be further expressed as the followings.

$$u = R_p \theta_p - R_g \theta_g - e + y_p - y_g \quad (2)$$

$$m_c = I_p I_g / (I_g R_p^2 + I_p R_g^2) \quad (3)$$

$$F_m = T_{pm} / R_p = T_{gm} / R_g \quad (4)$$

$$F_{aT}(t) = m_c R_p T_{pa}(t) / I_p \quad (5)$$

The nonlinear displacement function $f(u)$ can be described as the following

$$f(u) = \begin{cases} u - b, & u > b \\ 0, & -b \leq u \leq b \\ u + b, & u < -b \end{cases} \quad (6)$$

where $2b$ is the backlash value of the gear pair.

The periodic time-varying stiffness $k_m(t)$ can be given by [2]

$$k_m(t) = k_{mm} + \sum_{r=1}^n k_{ar} \cos(r\omega_m t + \varphi_{mr}) \quad (7)$$

where k_{mm} is the mean component of $k_m(t)$, k_{ar} is the amplitude of the r -th harmonic mesh stiffness content, φ_{mr} is the corresponding phase angle of the r -th harmonic mesh stiffness, n is the number of harmonic terms sufficient to describe the periodic function, ω_m is the fundamental frequency of the mesh stiffness.

Similarly, the static transmission error $e(t)$ can be expressed as

$$e(t) = \sum_{r=1}^n e_r \cos(r\omega_e t + \varphi_{er}) \quad (8)$$

where e_r is the amplitude of the r -th harmonic static transmission error, φ_{er} is the phase angle of the r -th harmonic static transmission error and ω_e is the fundamental frequency of static transmission error.

$F_{aT}(t)$ is assumed to include the same harmonics as $k_m(t)$ and $e(t)$ which can be formulated as

$$F_{aT}(t) = \sum_{r=1}^n F_{aTr} \cos(r\omega_{aT} t + \varphi_{aTr}) \quad (9)$$

where F_{aTr} is the amplitude of the r -th harmonic fluctuating force, φ_{aTr} is the phase angle of the r -th harmonic content and ω_{aT} is the fundamental frequency of input torque excitation.

The worn surfaces of gears can be regarded as faults deviating from the perfect gear surfaces, which in turn, will produce amplitude modulation (AM) and frequency modulation (FM) effects on gear mesh vibrations [34]. From this point of view, the mesh excitations arising from a gear pair with distributed worn surfaces can be modeled by amplitude and frequency modulations process [35]. For simplicity, only the fundamental harmonic of $k_m(t)$ and $e(t)$ are considered in this paper though the higher harmonics can be easily included. After considering the fundamental harmonic effect of surface wear the gear pair, one can formulate the mesh stiffness function with amplitude and frequency modulations as

$$k_m(t) = k_{mm} + a_k(t) k_{a1} \cos(\omega_m t + \varphi_{m1} + b_k(t)) \quad (10)$$

$$a_k(t) = 1 + A_k \cos(\omega_m t + \varphi_{ka}), \quad b_k(t) = B_k \sin(\omega_m t + \varphi_{kb}) \quad (11)$$

where A_k and B_k are the amplitudes of AM and FM, respectively; φ_{ka} and φ_{kb} are the initial phases of AM and FM, respectively.

Similarly, the static transmission error excitation with worn surfaces effects can be rewritten as the following

$$e(t) = a_e(t) e_1 \cos(\omega_e t + \varphi_{e1} + b_e(t)) \quad (12)$$

$$a_e(t) = 1 + A_e \cos(\omega_e t + \varphi_{ea}), \quad b_e(t) = B_e \sin(\omega_e t + \varphi_{eb}) \quad (13)$$

where A_e and B_e are the amplitudes of AM and FM, respectively; φ_{ea} and φ_{eb} are the initial phases of AM and FM, respectively.

The above modulation parameters can be pre-determined through loaded tooth contact analysis (LTCA) [32]. To ensure maximum portability and reusability, the LTCA was performed by using the commercial software Kisssoft®. In the LTCA, the gear tooth bending δ_{Zi} , the body deformation δ_{RKi} and the Hertz flattening $\delta_{H1,2}$ of an individual gear are taken into account ($i=p, g$) [36]. Therefore, the displacement of an individual gear along of the line of action can be summarized as

$$\delta_i = \delta_{Zi} + \delta_{RKi} + \delta_{H1,2} \quad (14)$$

The stiffness of one single tooth of an individual gear can be obtained as

$$k_i = \frac{F_i}{\delta_i} \quad (15)$$

Hence, the total stiffness of one tooth pair could be calculated as the following

$$k_{t1} = \frac{k_p + k_g}{k_p k_g} \quad (16)$$

In cases where there are two pairs in contact, the stiffness k_{t2} for the second tooth pair can be obtained with the same procedure. Then the equivalent mesh stiffness can be summarized as the follow

$$k_m = k_{t1} + k_{t2} \quad (17)$$

Accordingly, the static transmission error can be expressed as

$$e = \delta_p + \delta_g \quad (18)$$

The above procedure for determining the mesh stiffness and the static transmission error can be applied to a gear pair with and without worn surfaces.

After introducing the modulated excitations defined by Eq. (10) and (12) into Eq. (1), the nonlinear differential

equation of motion of the spur gear system can be solved by employing a fourth-order variable-step Runge-Kutta numerical integration method. With MATLAB coding, the dynamic behaviors such as steady-state dynamic responses of a spur gear system with worn surfaces can be determined, which is then integrated with the following quasi-static wear model to predict the dynamic wear of the gear system.

3. Dynamic wear prediction methodology for a spur gear system

In this section, a dynamic wear prediction methodology for spur gear system is presented by combining the aforementioned nonlinear dynamic model with a quasi-static wear model previously proposed by the authors [32]. For the convenience of understanding, the establishment of a quasi-static wear model for a spur gear system is briefly introduced though the details can be referred to our previous publication [32].

3.1. Quasi-static wear model for a spur gear system

The authors developed a quasi-static wear model to predict the surface wear of spur gears under static load conditions. The basic idea of this model is to employ Archard's wear equation to compute the wear depth of a local point on one of the mating surfaces in the integration form of the following

$$h = \int_0^s k p ds \quad (19)$$

where h is the wear depth, s is the sliding distance, p is the contact pressure and k is the wear coefficient.

Fig. 2 shows the flowchart of the overall wear prediction model.

As shown in Fig. 2, a three dimensional (3D) finite element (FE) model is developed to compute the mean value of contact pressures of meshed grids on the mating surfaces, which is then combined with the sliding distance and the wear coefficient to calculate the wear depth. Once the wear depth of any point on mating surfaces during one wear cycle is obtained, the above integral process is repeated until the wear depth of any point on the contact surfaces reaches a predetermined wear threshold, under which the contact pressures change accordingly. Then the gear surface needs to be reconstituted to perform another contact analysis for renewed surface pressures. The iterative process is repeated until the

maximum total wear depth on either of the two gears reaches a maximum allowable wear threshold. The wear depth of every point on the surface is obtained by summing up wear depths of every point accumulated in all processes of different pressure updates.

To be more specific, the wear depth occurs at each node ij of the fixed surface grid during one wear cycle can be calculated through Eq. (19) with calculated k , p and s . If the wear depth of any node on the contact surfaces reaches the predetermined wear threshold ε^q after the q -th pressure update, the geometry of gear surface needs to be reconstituted to perform contact analysis afresh for renewed surface pressure. The iterative process is repeated until the maximum total wear depth on either of the two gears reaches the maximum allowable wear threshold ε^t .

The accumulated wear depth at any node ij during one wear cycle is given as

$$\left(\Delta h_{ij}^q\right)_{\zeta}^{p,g} = k \left(\bar{p}_{ij}^q\right)_{\zeta}^{p,g} \left(s_{ij}^q\right)_{\zeta}^{p,g} \quad (20)$$

where $\left(\bar{p}_{ij}^q\right)_{\zeta}^{p,g}$ is the mean pressure value at node ij during one wear cycle, and $\left(s_{ij}^q\right)_{\zeta}^{p,g}$ is the sliding distance at node ij during one wear cycle after the q -th pressure update.

Eq. (20) is applied continuously ζ^q times until the wear depth accumulated at any node of either one of the contacting surfaces after the q -th geometry update reaches the threshold of ε^q . Then, the wear amounts at nodes ij accumulated since the last geometry update are given as:

$$\left(h_{ij}^q\right)_{\zeta}^{p,g} = \zeta^q \left(\Delta h_{ij}^q\right)_{\zeta}^{p,g} \quad (21)$$

The total number of wear cycles ζ^q is given as:

$$\zeta^q = \frac{\varepsilon^q}{\max\left\{\left(\Delta h_{ij}^q\right)_{\zeta}^p, \left(\Delta h_{ij}^q\right)_{\zeta}^g\right\}} \quad (22)$$

The wear amounts accumulated after each geometry updated are summed to get the final wear depth of node ij as:

$$h_{ij}^{p,g} = \sum_{q=1}^Q \left(h_{ij}^q\right)_{\zeta}^{p,g} \quad (23)$$

The total number of wear cycles required to reach to this wear depth is:

$$\zeta^t = \sum_{q=1}^Q \zeta^q \quad (24)$$

3.2. Dynamic wear prediction methodology

As can be observed from the above descriptions for quasi-static wear modeling, the computation for the surface wear of spur gears uses the mean values of contact pressures instead of dynamic contact loads on the mating surfaces. In a real-world gear system, however, the contact pressures on the mating surfaces are time-varying due to gear mesh excitations. Therefore, to predict the surface wear of a spur gear system under dynamic load conditions, we propose a dynamic wear model by integrating the nonlinear dynamic model with the quasi-static wear model. The computational methodology employed here to predict the dynamic wear of contacting gear surfaces is shown in Fig. 3.

As demonstrated in Fig. 3, an LTCA is performed at the beginning to obtain the static transmission error $e^q(t)$ and the time-varying stiffness $k_m^q(t)$ for each geometry update q . $e^q(t)$ and $k_m^q(t)$ are used as excitations for the nonlinear dynamic analysis to predict the steady-state dynamic responses of the gear system at a given rotational speed. As one of the response parameters, the relative displacement along the line of action after the q -th geometry update u^q is obtained and the dynamic gear mesh force (DMF) is calculated as $F_m^q(t) = k_m^q(t)f(u^q(t))$.

Based on the linear cumulative damage rule commonly referred as the Miner rule [37], the equivalent load T_{eq}^q is obtained from for each geometry update q [38]

$$T_{eq}^q = \left(\frac{n_1 (T_1^q)^l + n_2 (T_2^q)^l + \dots + n_j (T_j^q)^l}{n_1 + n_2 + \dots + n_j} \right)^{\frac{1}{l}} \quad (25)$$

where T_j^q is the torque at each discrete time instant t from $T_m^q(t)$, n_j is the corresponding times of occurrence of T_j^q , l is the slope of the Wöhler-damage line. And $T_m^q(t)$ can be calculated through the following equation

$$T_m^q(t) = R_p F_m^q(t) \quad (26)$$

The equivalent torque T_{eq}^q is input back into the contact model of the gear pair to compute the contact pressure p of engaging surfaces. By integrating the obtained contact pressure with sliding distance s and the wear coefficient k as formulated in Eq. (19), the wear depth of any point on the engaging surfaces can be computed. Similar to the descriptions for the flowchart demonstrated in Fig. 2, once the wear depth of any point on the contact surfaces reaches the

predetermined wear threshold to warrant a geometry update, the geometry of gear surface needs to be reconstituted to perform contact analysis afresh for renewed $e^g(t)$ and $k_m^g(t)$. The renewed dynamic analysis and contact analysis are utilized to obtain the updated pressures. With the updated pressures, the new wear computations are performed. The iterative process is repeated until the maximum total wear depth on either of the two gears reaches the maximum allowable wear threshold.

This study employs the same approximate method used by Ding and Kahraman [16] to take into account the influence of lubrication conditions at the gear mesh contact on the wear coefficient k . They expressed the wear coefficient k as

$$k = \begin{cases} k_0, & \lambda < \frac{1}{2} \\ \frac{2}{7}k_0(4 - \lambda), & \frac{1}{2} < \lambda < 4 \\ 0, & \lambda > 4 \end{cases} \quad (27)$$

where k_0 is the wear coefficient that corresponds to the values observed at relatively low speeds, λ is the so-called lambda ratio. The detailed descriptions can be found in the literature [16].

4. Numerical Simulation and Discussions

With the dynamic wear prediction methodology proposed in Section 3.2, a numerical example [5] is used to demonstrate the prediction of wear depth of engaging gear surfaces under dynamic condition. Then the coupling effects between the surface wear and the dynamic behaviors of the spur gear system are investigated. The design parameters are listed in Table 1. For comparing convenience, the investigations on the coupling effects are performed at the following operating speeds: (1) at $n_1=800$ rpm to represent low-speed dynamic conditions away from any resonance, (2) at $n_1=1100$ rpm near the second super-harmonic resonance peak, (3) at $n_1=1860$ rpm near the first super-harmonic resonance peak, (4) at $n_1=2200$ rpm to represent a higher speed off-resonance condition, (5) at $n_1=3720$ rpm near the primary resonance peak.

4.1. Effects of surface wear on mesh stiffness and static transmission error

The effects of surface wear on the mesh stiffness of the example system are demonstrated in Fig.4. As shown in Fig. 4(a), the spur gear pair experiences 1.74 mesh cycles from the starting point to the ending point. In general, the mesh

stiffness decreases with the increment of surface wear depths during the double contact zone. The influences of surface wear on the mesh stiffness are however, quite complicated during the transition region from double contact zone to single contact zone. This may be explained by the reason that the surface wear alters the contact condition of the gear pair.

Fig. 4(b) demonstrates the effects of maximum wear depth on the mean value and the root-mean-square (rms) value as well as the first three harmonic amplitudes of the mesh stiffness of the example system. As can be clearly observed, the mean value of the mesh stiffness k_{mm} decreases monotonously with the accumulation of wear depth. Meanwhile, the rms value and the first three harmonic amplitudes of the mesh stiffness almost monotonously increase with the accumulated surface wear. Compared to the first harmonic amplitude k_1 and the third harmonic amplitude k_3 , variations of the rms value k_{rms} and the second harmonic amplitude k_2 caused by the surface wear are quite noticeable. The wear-caused variations of these harmonic components of mesh stiffness will aggravate the internal excitation of time-varying mesh stiffness, which in turn, degenerates the dynamics behaviors of the gear system.

The effects of surface wear on the STE of the example system are demonstrated in Fig. 5. As can be observed from Fig. 5(a), the STE decreases monotonously with the increment of wear depth during the majority of the mesh cycles. Similar to the variations of mesh stiffness, the influences of wear depth on the STE are complicated at the transition region of double contact zone to single contact zone. Fig. 5(b) shows that the influences of wear depth on the rms value $(\text{STE})_{\text{rms}}$ and the second harmonic amplitude $(\text{STE})_2$ are much more ‘stronger’ than those on $(\text{STE})_1$ and $(\text{STE})_3$. As one of the major internal excitation sources, the variations of STE caused by surface wear will also change the dynamic characteristics of the gear system.

Time-varying mesh stiffness excitation and gear transmission error excitation are the main contributors to gear vibration and noise. Accordingly, the influences of surface wear on mesh stiffness and transmission error must be considered in the problem of gear vibration and noise.

4.2. Effect of surface wear on dynamic characteristics

By considering the internal excitation contributions from the surface wear, the dynamic behaviors of the example system are predicted with the nonlinear dynamic model. In order to evaluate the effects of surface wear on the system dynamics, the dynamic mesh force F_m of the spur gear pair without surface wear is taken as the baseline at each operational speed condition mentioned above.

As shown in Fig. 6, F_m fluctuates within a certain range during the mesh cycles. Compared with high speed or resonant operational conditions, the fluctuation of F_m is smaller when the system works at lower speed or non-resonant situation. For low or medium speed operations (the first four operational conditions), F_m is always a non-zero value, demonstrating that there barely exists tooth separations. On the contrary, when the system works at high speed of $n_1=3720$ rpm, the dynamic mesh force F_m becomes zero for about 44% of the time domain caused by tooth separations. A typical circumstance occurs when the system works at the condition of $n_1=1860$ rpm in that F_m almost turns to be zero at one position during a whole mesh period. It implies that the gear system reaches the ‘threshold’ of tooth separations at this time interval.

By comparing the above dynamic mesh force F_m with the results in the study of Ding and Kahraman [16], we can find that the steady-state dynamic responses obtained from the present model and Ding’s model are slightly different. For example, the value of F_m reaches zero twice during a mesh period when the system works at $n_1=1860$ rpm and F_m becomes zero for about 40% of time domain when the operating speed is $n_1=3720$ rpm in Ding’s study [16]. The differences in dynamic responses are aroused by the translational deflections that are considered in our model. With the consideration for additional translational motions of the gears, the prediction accuracy for nonlinear dynamic behaviors of the system can be improved.

Fig.7 shows the time histories of F_m at the five operation speeds after the gear system runs for 3, 20, 40 and 100 million cycles of wear. By comparing Fig. 7 with Fig. 6, the effects of surface wear on dynamic responses of the example system can be revealed.

In general, the peak-to-peak amplitude of F_m first decreases and then increases with the increment of wear cycles. Meanwhile, the tooth separations can be observed after some wear cycles. However, the effects of surface wear on F_m

are not exactly the same for different operational conditions. To be specific, it can be observed from Fig. 7(a) that the peak-to-peak amplitude of F_m at $n_1=800$ rpm firstly decreases and then increases after 20 million wear cycles; meanwhile, obvious tooth separations can be observed after 100 million wear cycles though the separation duration in each mesh cycle is very short.

Fig. 7(b) shows that the peak-to-peak amplitude of F_m at $n_1=1100$ rpm firstly decreases with wear and then increases after 40 million wear cycles; meanwhile, obvious tooth separations occur at 100 million wear cycles though the separation duration in each mesh cycles is also very short.

Similarly, the peak-to-peak amplitude of F_m at $n_1=1860$ rpm decreases slightly and then increases at the tuning point of 20 million mesh cycles. The tooth separations take place after 40 million wear cycles and become obvious after 100 million wear cycles. The peak-to-peak amplitude of F_m at $n_1=2200$ rpm decreases slightly before 20 million wear cycles and then increases dramatically after 100 million wear cycles. Meanwhile, the tooth separations last for about 44% of one mesh cycle after 100 million wear cycles.

It is very interesting that the tooth separations occur in the unworn gear pair worked at $n_1=3720$ rpm (as observed in Fig. 6) are eliminated after 3 million wear cycles and reappear after 100 million wear cycles as shown in Fig.7(e). Moreover, the peak-to-peak amplitude of F_m increases monotonously with wear cycles, which is different from those of low and medium operation speeds. It can be roughly concluded that the initial surface wear may contribute to the reduction of the dynamic mesh force amplitudes and also help to eliminate tooth separations of the gear pair when the system operates at a resonance peak. After some wear cycles, the accumulated wear amounts may increase the amplitudes of dynamic mesh force and arouse tooth separations again. The influences of surface wear on the nonlinear dynamic behavior are more outstanding when the gear system operated at a higher speed.

The effects of surface wear on equivalent load of the spur gear system are further investigated and depicted in Fig. 8. For a spur gear system without worn surfaces, the equivalent load at primary resonance peak ($n_1=3720$ rpm) is much larger than those at the other speeds. However, situations may change after some wear cycles. For example, the equivalent load at an off-resonance speed of $n_1=2220$ rpm claims the highest value after the gear system runs for 100 million wear cycles.

Similar to the variations of dynamic mesh forces, the equivalent load firstly decreases and then increases with the increment of wear cycles at the five speeds.

To address the influences of surface wear more clearly, this paper defines the dynamic factor of the gear pair as $DF=(DMF)_{\max}/SMF$, where $(DMF)_{\max}$ is the maximum value of the dynamic mesh force in a mesh cycle and SMF is the static mesh force transmitted by the gear pair. The influences of surface wear on the dynamic factor are shown in Fig. 9.

As can be clearly observed, DF achieves the highest value at $n_1=3720$ rpm for a gear system without surface wear. This DF value drops dramatically after 3 million wear cycles and then increases monotonously with the wear accumulations. Similar trends can also be found in the other four operation conditions except that the DF values are smaller and the turning points of wear cycles are different. Once again, it can be found that the influences of surface wear on DF become more obvious when the gear system working at high speed ranges. Also, it can be concluded that accumulated wear depth may aggravate the system dynamic behaviors.

The following will explore the effects of surface wear on DTE of the spur gear pair. The simulation results are shown in Fig. 10.

As can be observed from Fig. 10, the root-mean-square value of the dynamic transmission error $(DTE)_{\text{rms}}$ decreases with mild wear of the gear surfaces and then increases with the accumulation of surface wear when the system works at low operating speed of $n_1=800$ rpm and high operating speed of $n_1=3720$ rpm. On the contrary, $(DTE)_{\text{rms}}$ decreases with the wear cycles at the operating speeds of $n_1=1100$ rpm, $n_1=1860$ rpm and $n_1=2200$ rpm.

At low-speed conditions away from any resonance ($n_1=800$ rpm), the third harmonic of DTE $(DTE)_3$ is higher than the first harmonic of DTE $(DTE)_1$ and the second harmonic of DTE $(DTE)_2$, making $(DTE)_{\text{rms}}$ is dominated by $(DTE)_3$. Surface wear is capable of increasing $(DTE)_3$ resulting in an increase in $(DTE)_{\text{rms}}$ beyond 20 million wear cycles. Near the superharmonic resonances ($n_1=1100$ rpm and $n_1=1860$ rpm), $(DTE)_2$ is higher and dominates $(DTE)_{\text{rms}}$. $(DTE)_{\text{rms}}$ decreases with wear cycles duo to the reduction of $(DTE)_2$ beyond 40 million cycles. At higher speed off-resonance conditions ($n_1=2200$ rpm), $(DTE)_3$ is very small compared to $(DTE)_1$ and $(DTE)_2$. $(DTE)_{\text{rms}}$ is primarily impacted by $(DTE)_1$ and $(DTE)_2$. When the operating speed approaches the primary resonance peak ($n_1=3720$ rpm), $(DTE)_2$ and $(DTE)_3$ are smaller

than $(DTE)_1$ and as a result $(DTE)_{rms}$ is dominated by $(DTE)_1$. The first three harmonic amplitudes of DTE increase with wear significantly beyond 40 million cycles, making $(DTE)_{rms}$ turn to be higher.

Fig. 11 shows the effects of surface wear on the period of steady dynamic response of the example system.

As can be seen from Fig. 11, the periods of steady dynamic response retain the same after different wear cycles when the gear system operates at the other four speeds except $n_1=3720$ rpm. The period-four, period-three, period-two and period-eight responses are successively observed in dynamic analyses when the wear cycles increase from 0 to 100 million.

4.3. Effect of gear dynamics on surface wear

In order to study the effect of gear dynamics on surface wear, the static wear behaviors are adopted as a baseline for comparisons. The surface wear distributions of the pinion after different wear cycles under static load conditions are depicted in Fig. 12.

Fig. 13 indicates that during the early stages of wear, the influence of operational conditions on the maximum wear depth is quite trivial. The gaps of accumulated wear amounts between the five operational conditions and the static condition are very small. However, the accumulated wear depths change significantly with the operational speed when wear cycles exceed a range. The maximum wear depths at the four operation speeds (except for $n_1=3720$ rpm) are larger than that of the static condition after the same wear cycles. On the contrary, the maximum wear depth at the primary resonant speed of $n_1=3720$ rpm is smaller than that of the static condition. Besides, the maximum wear depth accumulated at $n_1=1100$ rpm is the largest and the maximum wear depth at $n_1=3720$ rpm is the smallest. However, there exists no simple linear trend between the maximum wear depth and the operating speed. For example, the maximum wear depths at $n_1=1860$ rpm and $n_1=2200$ rpm are smaller than those at $n_1=1100$ rpm. The reason may lie in that the accumulated wear depth is affected by both the contact pressures and the wear coefficient. And the contact pressures are related to the mesh forces, which changes with the system's operating speed. The wear coefficient decreases when the operating speed increases [16]. When the gear system operates at $n_1=1100$ rpm, the equivalent load and the wear

coefficient are larger than those at other speeds, thus leads to the largest wear depth value at this operational condition.

While at the primary resonance peak ($n_1=3720$ rpm), due to the tooth separations, the contact chances of engaged mating surfaces are reduced to slow the accumulation of surface wear. To summarize, the wear depth is the largest when the example system operates near the second super-harmonic resonance peak while achieves the smallest value near the primary resonance peak. The wear depth near a low-speed off-resonance condition is significantly smaller than those near the second super-harmonic resonance peak and the first super-harmonic resonance peak. The wear depth value near a higher speed off-resonance condition is larger than those near a low-speed off-resonance condition.

5. Conclusions

A dynamic wear prediction methodology is proposed by combining a translational-rotational-coupled nonlinear dynamic model with a quasi-static wear model to investigate the coupling effects between surface wear and dynamic behaviors in a spur gear system. Based on the analyses carried out in this study, the following conclusions can be drawn:

(1) The accumulated surface wear alters the mesh stiffness and the transmission error of the gear pair, thus affects the dynamic characteristic of the gear system significantly.

(2) The initial surface wear contributes to reduce the amplitudes of dynamic response and eliminate tooth separations of the gear pair operated at a resonance peak. However, additional wear amounts may aggravate the system vibration characteristics and cause tooth separations.

(3) The effects of surface wear on period of steady dynamic response are negligible except for the primary resonance peak.

(4) To improve the wear prediction accuracy, the translational deflections in the gear system must be included.

(5) During the mild wear stages, the effect of operational condition on surface wear is very weak. However, the wear depths are affected by the operating speed noticeably after some wear cycles.

(6) The wear depth values are the largest when the gear system operates near the second super-harmonic resonance peak while achieve the smallest when operates near the primary resonance peak.

Our future work focuses on the effect of load, tooth modification, bearing stiffness and misalignment on both system dynamics and surface wear.

Acknowledgements

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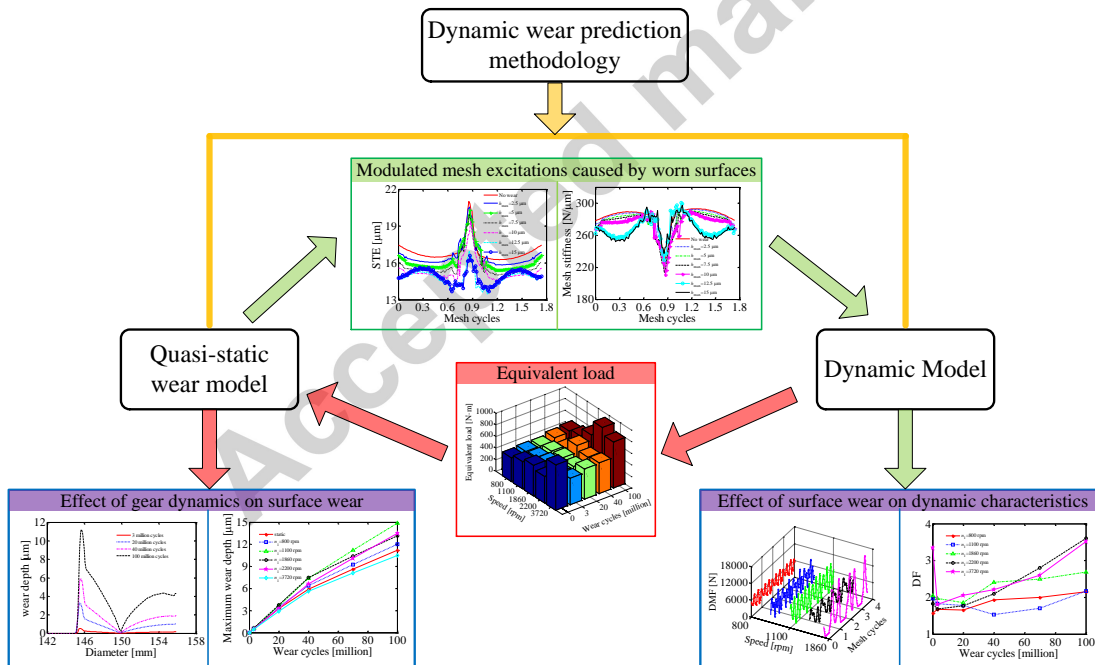
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Highlights

1. A combined model based dynamic wear prediction methodology for a spur gear system is proposed.
2. The proposed dynamic model takes into account the translational deflections in the gear system to improve the wear prediction accuracy.
3. The coupling effects between surface wear and dynamic behaviors in a spur gear system are investigated quantitatively.

Graphical abstract



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Fig. 11 Phase plane plots of steady dynamic response u after various wear cycles at (a) $n_1=800$ rpm, (b) $n_1=1100$ rpm, (c)

$n_1=1860$ rpm, (d) $n_1=2200$ rpm, (e) $n_1=3720$ rpm

Fig. 12 Wear distributions of the pinion after various wear cycles at static operation condition

Fig. 13 Variation of maximum wear depth with respect to wear cycles at different operational conditions

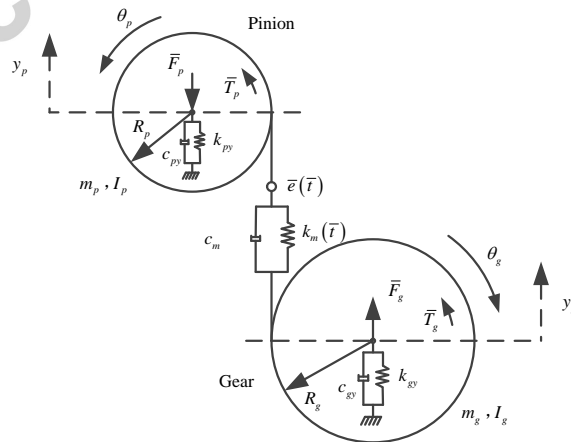


Fig. 1. Nonlinear dynamic model for a spur gear system.

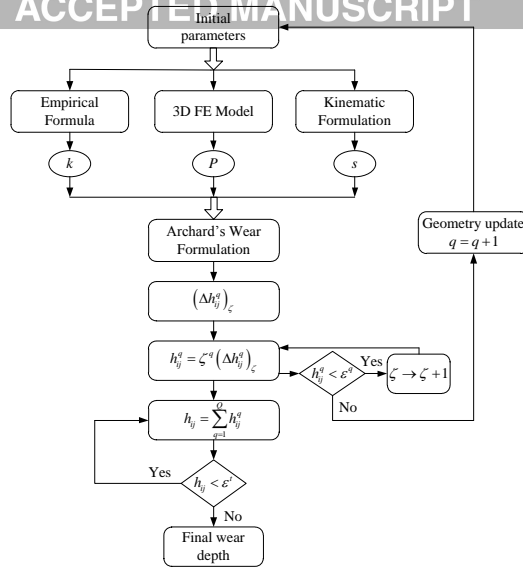


Fig. 2. Computational methodology of gear wear.

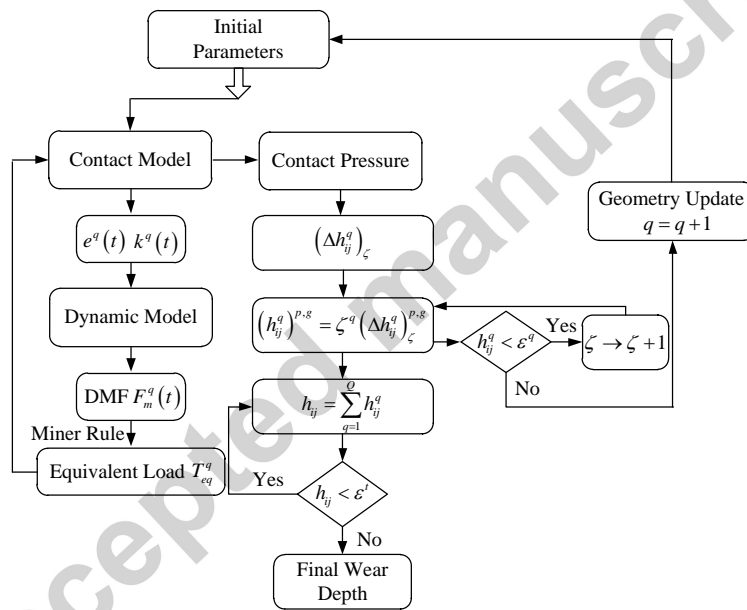


Fig. 3. Computational methodology of dynamic wear.

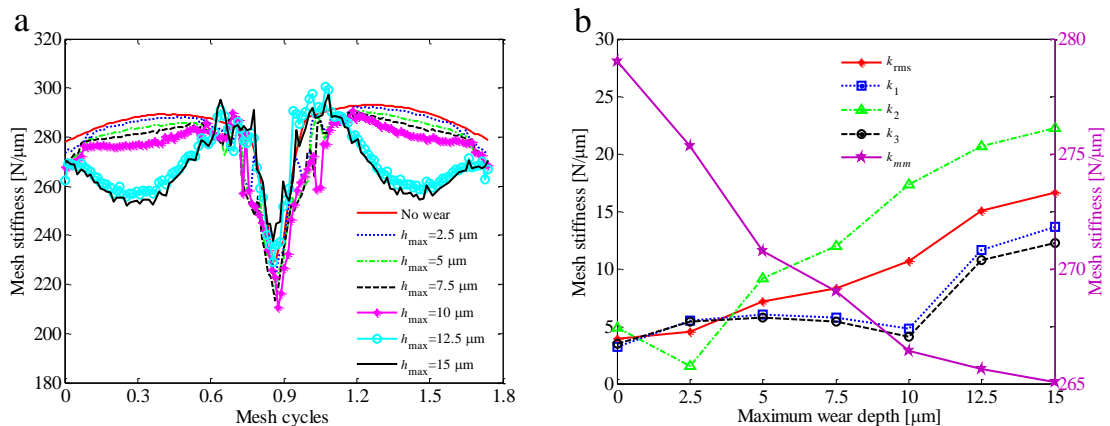


Fig. 4. Variation of mesh stiffness with maximum wear depth.

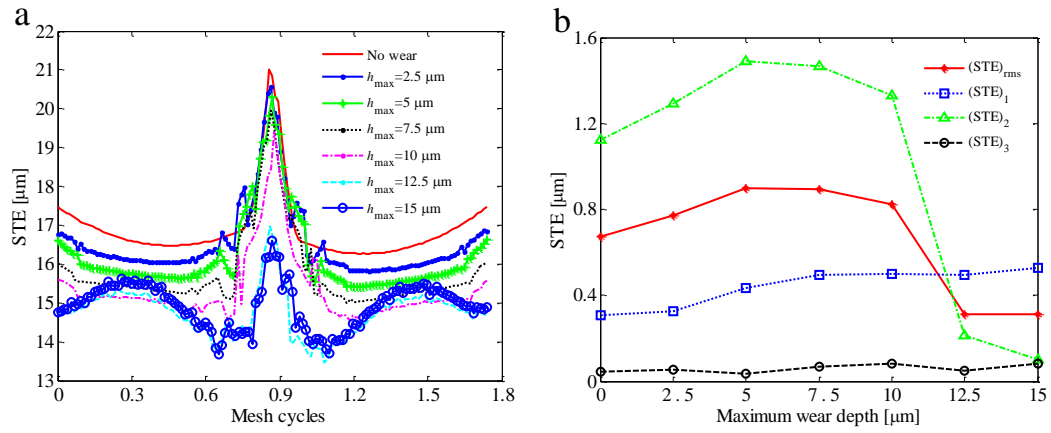
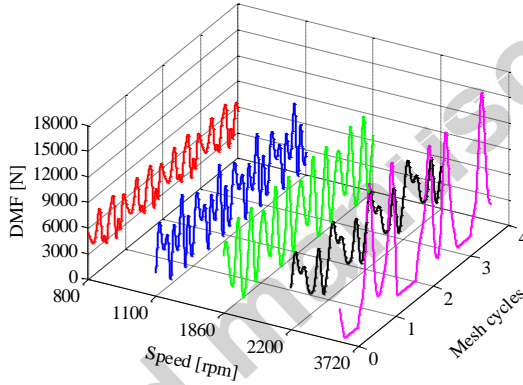
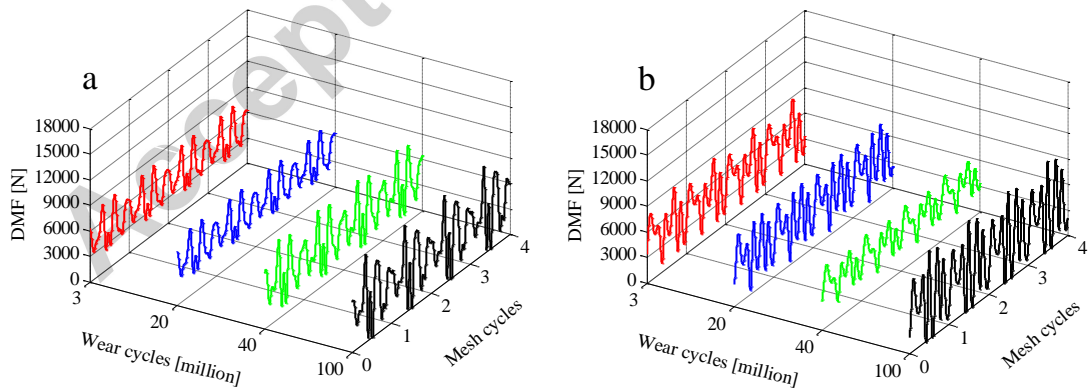


Fig. 5. Variation of static transmission error with maximum wear depth.

Fig. 6. Time histories of F_m without wear at five operational conditions.

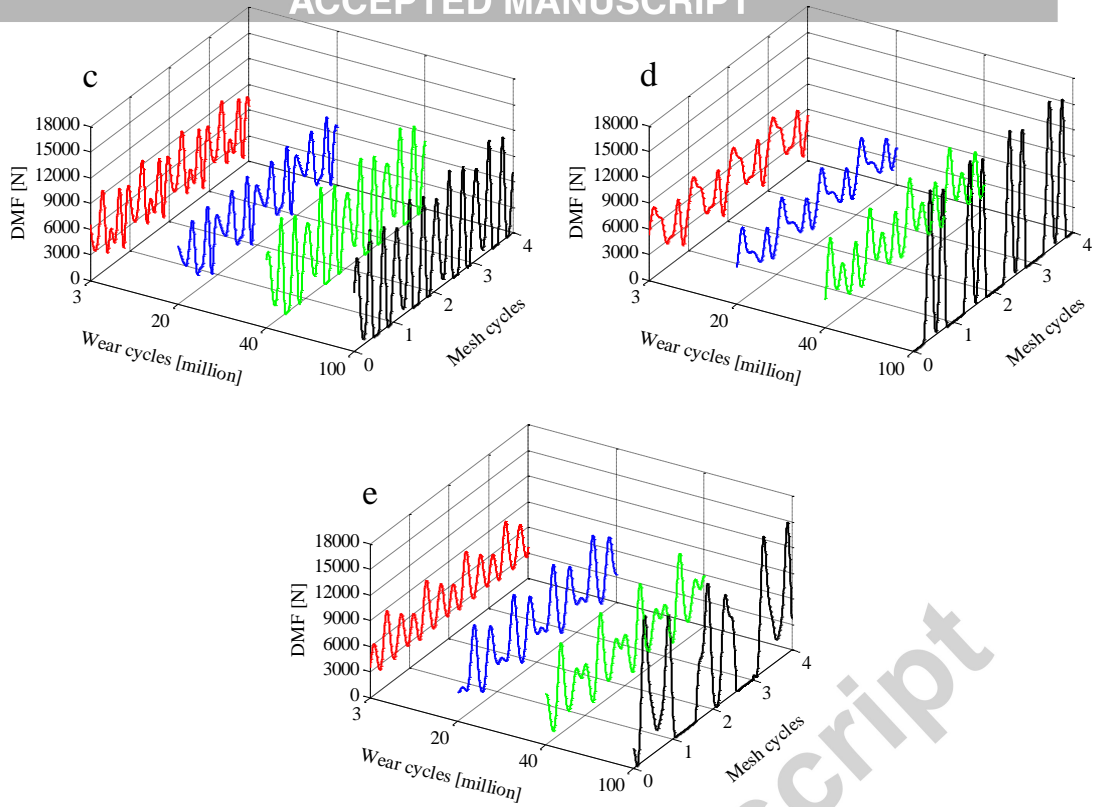


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(e) $n_1=3720$ rpm.

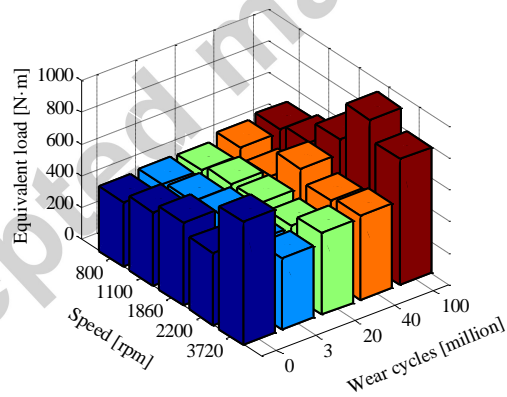


Fig. 8. Variations of equivalent load with wear cycles.

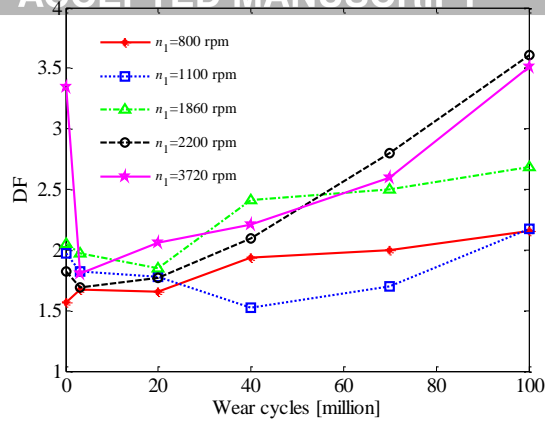
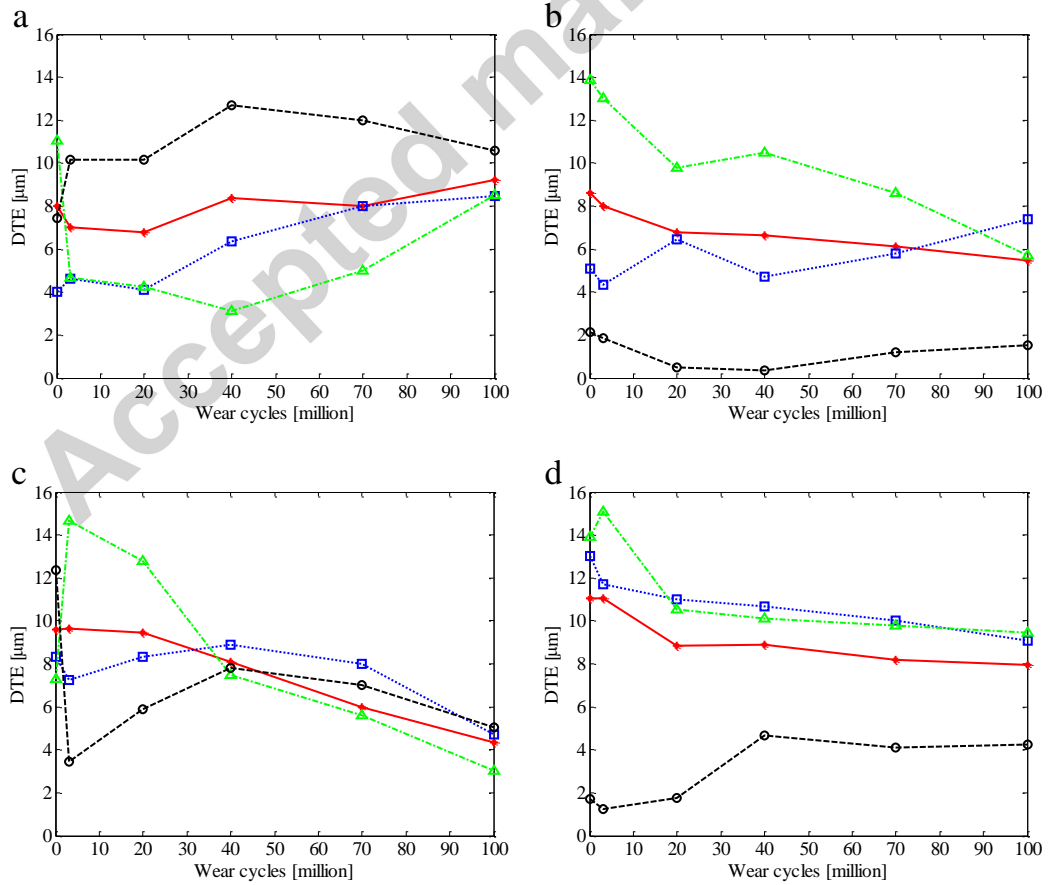


Fig. 9. Variations of dynamic factor with wear cycles.



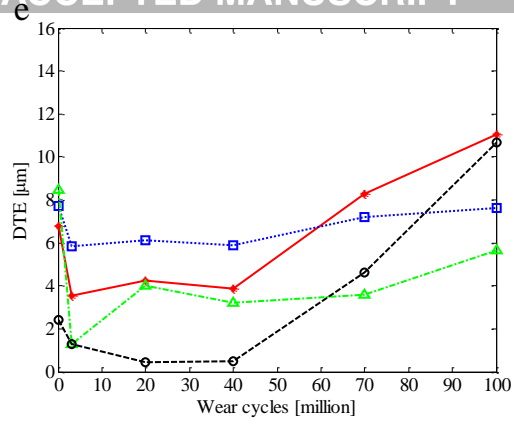
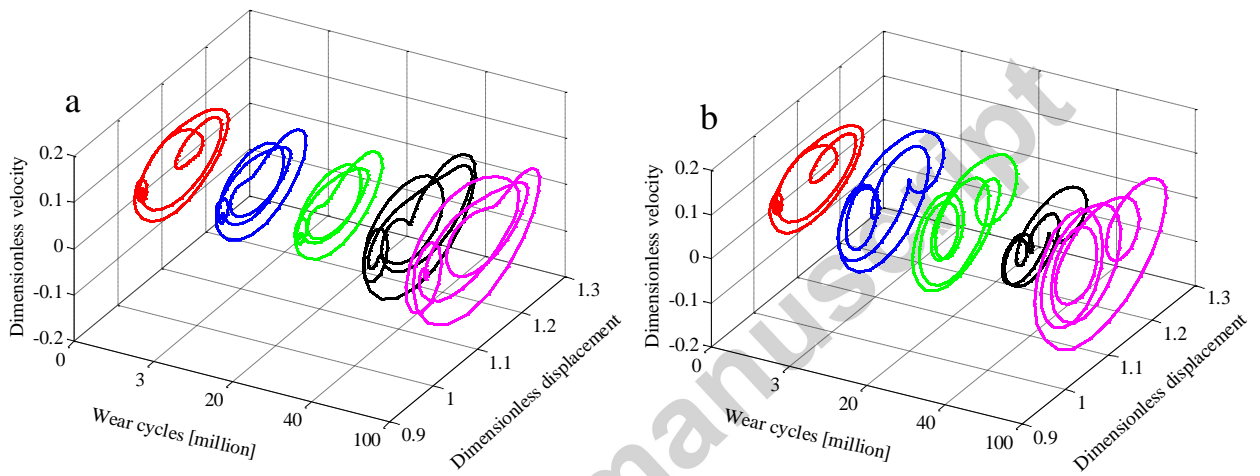


Fig. 10. Variation of dynamic transmission error with wear cycles at (a) $n_1=800$ rpm, (b) $n_1=1100$ rpm, (c) $n_1=1860$ rpm, (d) $n_1=2200$ rpm, (e) $n_1=3720$ rpm. (—*) (DTE)_{rms}, (····) (DTE)₁, (—Δ—) (DTE)₂, (—○—) (DTE)₃.



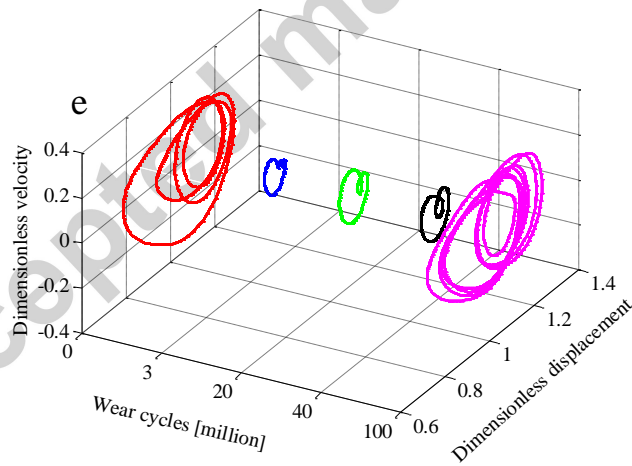
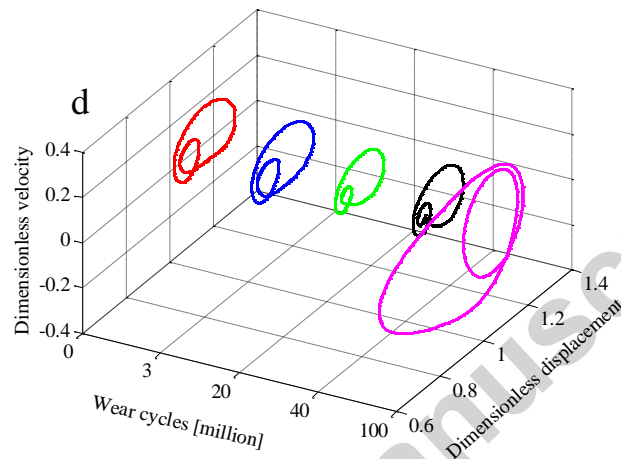
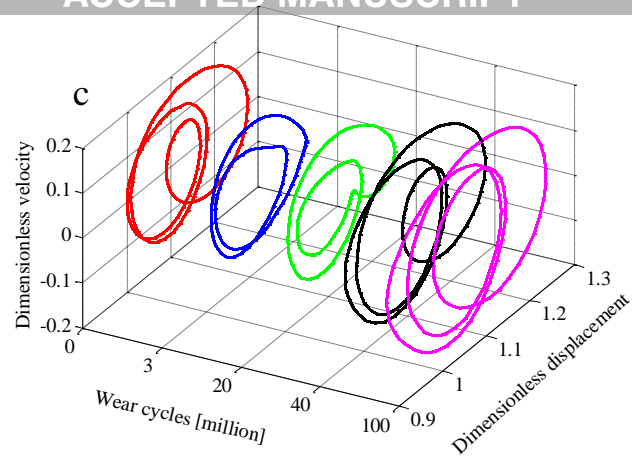


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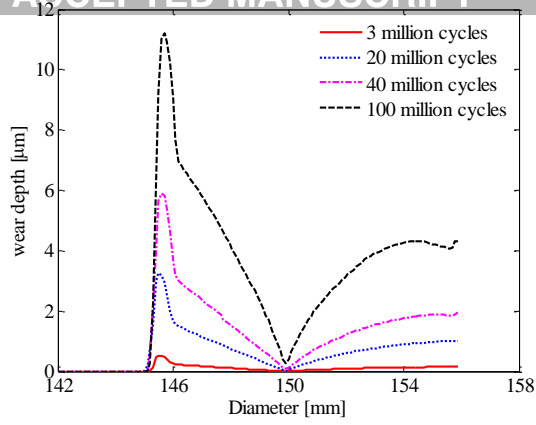


Fig. 12. Wear distributions of the pinion after various wear cycles at static operation condition.

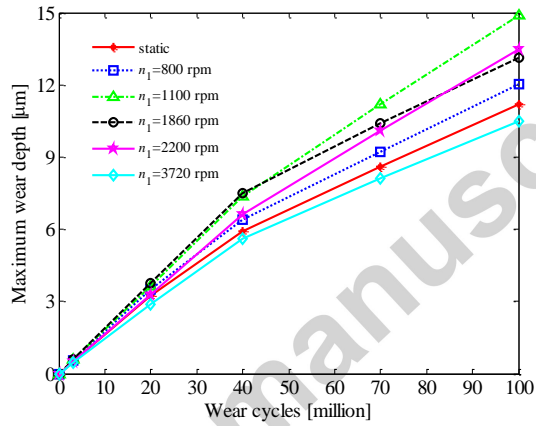


Fig. 13. Variation of maximum wear depth with respect to wear cycles at different operational conditions.

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Table 1 Parameters of the example system.

Table 1 Parameters of the example system.

Parameter Nomenclatures	Pinion p	Gear g	Parameter Nomenclatures	Pinion p	Gear g
Number of tooth z	50	50	Module m_n /mm	3.0	3.0
Pressure angle α_n (°)	20	20	Tooth width B /mm	20	20
Pitch diameter d_w /mm	150	150	Base diameter d_b /mm	140.95	140.95
Addendum diameter d_a /mm	156	156	Dedendum diameter d_f /mm	140	140
Material	42CrMo	42CrMo	Density ρ /(kg·m ⁻³)	7830	7830
Young's modulus E /GPa	206	206	Poisson's ratio ν	0.3	0.3

ACCEPTED MANUSCRIPT					
Mass m /kg	2.8	2.8	Moment of inertia I /(kg·mm ²)	7300	7300
Bearing stiffness k_b /(N/μm)	1150	1150	Bearing damping c /(Ns/m)	5360	5360
Roughness R_a /μm	0.3	0.3	ISO precision grade	5	5
Meshing damping c_m /(N·s/m)				1000	
Input torque T_p /(N·m)				340	
Center distance a /mm				150	

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